Twin Otter SERIES 400

Airframe Technical Record

Volume Number: 1 **MSN: 976**

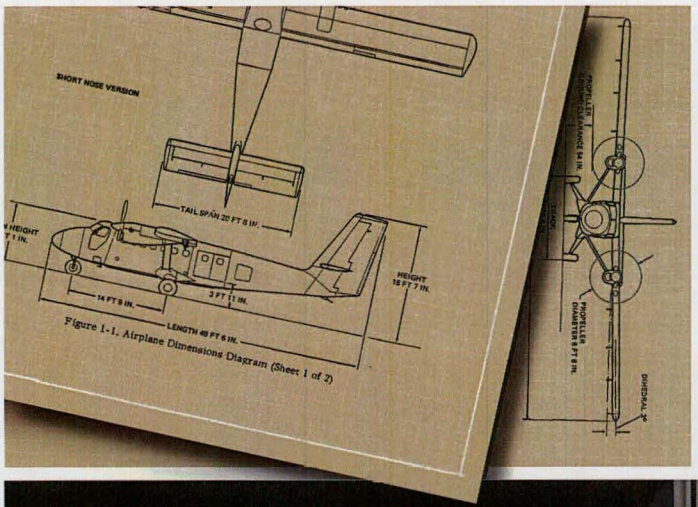


Figure 1-1. Airplane Dimensions Diagram (Sheet 1 of 2)





VIKING

VERSATILITY THAT WORKS

Airframe Data

Nationality & Registration: **C-GYVA**

Manufacturer's Designation: **DHC-6 Series 400**

Manufacturer's Serial Number: **976**

Type Certificate Number: **A-82**

Volume Number: **1**

Date Opened: **26-Mar-2019**

Airframe Technical Record Contents

Section 1 – Airframe Technical History

Section 2 – Major Component Installations

Section 3 – Airworthiness Directives

Section 4 – Mandatory Service Bulletins

Section 5 – Modification Status

Definitions

For the purposes of this technical record the following definitions apply:

TTSN - Total Time Since New
TCSN - Total Cycles Since New
MSN - Manufacturer's Serial Number
UTC - Coordinated Universal Time
FAA - Federal Aviation Administration
EASA - European Aviation Safety Agency
TCCA - Transport Canada Civil Aviation Authority
EMMA - Equalized Maintenance for Maximum Availability

MOD - Modification
SOO - Standard Order Option
AD - Airworthiness Directive
SB - Service Bulletin
TB - Technical Bulletin
STC - Supplemental Type Certificate
LSTC - Limited Supplemental Type Certificate
RDIMS - Records, Document and Information Management System

Airtime - The time from the moment an aircraft *leaves the surface* until it comes into contact with the surface at the next point of landing.

Engine Cycle - 1 Cycle = Start-up + Take-off + Landing + Shutdown

Landing Cycle - 1 Cycle = Take-off + Landing

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carry forward to the top of next page ▶▶▶▶	976	0.0	0	0	0

Production Activities

Initial Production Release

This aircraft has been manufactured in accordance with Viking Air Limited production system as per the latest revision of engineering drawing C6-1000. All applicable ground testing has been completed and this aircraft is considered ready for production flight testing.

Pursuant to C.A.R. Part VI, subpart 5, section 605.86, this aircraft shall be maintained in accordance with the EMMA Controlled Inspection Program. The Pre-Flight Inspection Checklist (ref. FORM-017-27) satisfies the EMMA 7 day inspection cards numbered Routine A and Routine E. The Pre-Flight Inspection Checklist shall be conducted prior to the first flight of each day.

A flight authority has been issued by Transport Canada for this aircraft, as per CARs, Part V, Subpart 7, section 507.04.

RDIMS::
14804398

PRIOR TO EACH flight the operating limitations shall be determined and annotated in the appropriate location by authorized personnel. The following must be referenced:

- Appendix A – Production Test Flight
- Appendix B – Ferry Flights
- Appendix C – Customer Acceptance Flights

If the flight authority described above is not used it shall be annotated as "other" in the appropriate location along with the Transport Canada issued RDIMS number in the space provided.

Pursuant to C.A.R. Part II, Subpart 2, Section 202.07, all required placards and marking have been affixed to the aircraft. This aircraft has been registered with the following marks:

Registration Marks:
C-GYVA

The aircraft engines, propellers and flight controls have been rigged as per the latest revision of engineering drawing C6-1006.
The following propeller blade angle settings have been configured as per the latest revision of the applicable propeller installation drawing:

Feather Pitch:	Start Lock Pitch:	Low Pitch (Prime):	Low Pitch (Beta Pick-up):	Reverse Pitch:
+87.0° (± 1.0°)	N/A	+17.0° (± 0.5°)	+21.0° (±0.5°)	-15.0° (± 0.5°)

The following aircraft systems have been tested as per the latest revision of the applicable ground test procedures, and are considered to meet the applicable Canadian Airworthiness Requirements:

- Electronic Flight Instrument System
- Transponders
- Attitude and Heading Reference System (ADAHRS)
- Emergency Standby Instrument System (ESIS)
- Emergency Locator Transmitter (ELT)

Certification Statement

As of this date, I certify that this Aircraft, complies with all applicable Airworthiness Directives.

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	DEC 1 3 2018	VA MFG 1568

The above entry has been transcribed from the Aircraft Journey Logbook.	Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	0.0	0	0	0

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☐ Application of Exterior Paint ☐ Interior Completion ☑ Other Flight Test Configuration

Basic Aircraft Weight (lbs):	Center of Gravity (inches):
6872	211.71

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:
	DEC 17 2018	VA MFG 5009

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: DEC 17 2018
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2018-34
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☑ Appendix A (Production Flight Testing) ☐ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☐ Other _____	RDIMS: 14804398

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	DEC 17 2018	VA MFG 1568

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):
CYYC	CYYC	7385	2305

Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
1923	2154	2.5	1	1	1

Carry forward to the top of next page ▶▶▶▶

New TTSN	New TCSN		
	#1 Engine	#2 Engine	Landings
2.5	1	1	1

Pilot In Command

Signature/Name:	Date:	License Number:
Taylor Williams	17 Dec 2018	AA 500840

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	2.5	1	1	1

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction ~~WI-017-14~~ following:

☐ Application of Exterior Paint ☐ Interior Completion ☐ ~~Other~~ _____

Basic Aircraft Weight (lbs):	Center of Gravity (inches):

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, ~~the presence of all~~ safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date: — Date: DEC 17 2018

As of this date all applicable Canadian Airworthiness Directives have been complied with up to: — TCCA AD#: CF-2018-34

This aircraft is released under the authority of the following Specific Purpose Flight Permit: — RDIMS: 14804398

☑ Appendix A (Production Flight Testing) ☐ Appendix B (Ferry Flights)
☐ Appendix C (Customer Acceptance) ☐ Other _____

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	DEC 17 2018	VA MFG 1568

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A) — Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):
CYYC	CYYC	7885	2480

Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
2300	2327	0.5	1	1	1

Carry forward to the top of next page ▶▶▶▶	New TTSN	New TCSN		
		#1 Engine	#2 Engine	Landings
	3.0	2	2	2

Pilot In Command

Signature/Name:	Date:	License Number:
Taylor Williams	17 Dec 2018	AA[illegible]40

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	3.0	2	2	2

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☐ Application of Exterior Paint ☐ Interior Completion ☑ Other CVR Removed

Basic Aircraft Weight (lbs):	Center of Gravity (inches):
6862.80	211.44

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:
	JAN 0 2 2019	VA MFG 5009

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: JAN 0 2 2019
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2018-36
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☑ Appendix A (Production Flight Testing) ☐ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☐ Other	RDIMS: 14804398

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	JAN 0 2 2019	VA MFG 1568

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):
CYYC	CYYC	7855	2295

Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
1811	1837	0.3	1	1	1

Carry forward to the top of next page ▶▶▶▶	New TTSN	New TCSN		
		#1 Engine	#2 Engine	Landings
	3.3	3	3	3

Pilot In Command

Signature/Name:	Date:	License Number:
DENKO	02 JAN 2019	AA295329

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	3.3	3	3	3

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☐ Application of Exterior Paint ☐ Interior Completion ☐ Other _____

Basic Aircraft Weight (lbs):	Center of Gravity (inches):

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: JAN 0 2 2019
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2018-36
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☐ Appendix A (Production Flight Testing) ☑ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☐ Other _____	RDIMS: 14804398

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	JAN 0 2 2019	VA MFG 1568

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date: N/A

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):
CYYC	KGEG	7355	2410

Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
2007	2157	1.8	1	1	1

Carry forward to the top of next page ▶▶▶▶	New TTSN	New TCSN		
		#1 Engine	#2 Engine	Landings
	5.1	4	4	4

Pilot In Command

Signature/Name:	Date:	License Number:
X Mark Goodrich	02 JAN 2019	AF79857

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	5.1	4	4	4

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☑ Application of Exterior Paint ☐ Interior Completion ☐ Other _____

Basic Aircraft Weight (lbs):	Center of Gravity (inches):
6969.80	~~212~~ 212.61

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:
	JAN 31 2019	VA MFG 1503

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: JAN 31 2019
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2018-36
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☐ Appendix A (Production Flight Testing) ☑ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☐ Other _____	RDIMS: 14804398
The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.	

Signature:	Date:	Stamp:
	JAN 31 2019	25 QMQC

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):
KGBG	CYYJ	7480	2440

Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
2300	0054	1.9	1	1	1

	New TTSN	New TCSN		
		#1 Engine	#2 Engine	Landings
Carry forward to the top of next page ▶▶▶▶	7.0	5	5	5

Pilot In Command

Signature/Name:	Date:	License Number:
D. MONTGOMERY	31 JAN 2019	A799307

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	7.0	5	5	5

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☐ Application of Exterior Paint ☑ Interior Completion ☑ Other INSTALL CVR

Basic Aircraft Weight (lbs):	Center of Gravity (inches):
7204.00	213.01

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:
	MAR 13 2019	VA MFG 1503

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: MAR 13 2019
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2019-09
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☐ Appendix A (Production Flight Testing) ☐ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☑ Other N/A	RDIMS: 14804398

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	MAR 13 2019	VA MFG 798

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):		
Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
Carry forward to the top of next page ▶▶▶▶		New TTSN	New TCSN		
			#1 Engine	#2 Engine	Landings

Pilot In Command

Signature/Name:	Date:	License Number:

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ►►►►	976	7.0	5	5	5

Production Activities

Weight and Balance

A weight and balance has been performed/amended in accordance with Viking Air Limited work instruction WI-017-14 following:

☐ Application of Exterior Paint ☐ Interior Completion ☐ Other _____

Basic Aircraft Weight (lbs):	Center of Gravity (inches):

Independent Inspection

I have performed an independent inspection of the affected engine and/or flight control systems for correct assembly, the presence of all safety locking devices, and the correct sense of movement through the full range of motion.

Signature:	Date:	Stamp:
	MAR 2 0 2019	VA DMFL 798

Production Release

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date:	Date: MAR 2 0 2019
As of this date all applicable Canadian Airworthiness Directives have been complied with up to:	TCCA AD#: CF-2019-10
This aircraft is released under the authority of the following Specific Purpose Flight Permit: ☐ Appendix A (Production Flight Testing) ☐ Appendix B (Ferry Flights) ☐ Appendix C (Customer Acceptance) ☑ Other N/A	RDIMS: 14804398

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	MAR 2 0 2019	VA DMFL 798

Flight Activities

Flight Details

Pre-Flight Inspection Checklist (FORM-017-27) must be completed day of flight. If Production Release is valid, inspection may be performed by Flight Crew. Enter date inspected by Flight Crew (otherwise N/A)

Date:

Point of Departure:	Destination:	Operating Weight (lbs):	Fuel Weight (lbs):		
Up Time (UTC hh:mm):	Down Time (UTC hh:mm):	Air time this flight (6 min equals 0.1 hr)	Cycles this flight		
			#1 Engine	#2 Engine	Landings
Carry forward to the top of next page ►►►►		New TTSN	New TCSN		
			#1 Engine	#2 Engine	Landings

Pilot In Command

Signature/Name:	Date:	License Number:

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
Carried forward from the previous page ►►►►	976	7	#1 Engine 5	#2 Engine 5	Landings 5

Production Activities

Final Production Statement

Magnetic Heading Reference Calibration has been completed in accordance with the applicable ground test procedures:

Calibration Date		06-Dec-2018										
Component	360	030	060	090	120	150	180	210	240	270	300	330
ADAHRS A	360	029	059	089	120	150	179	209	240	271	300	330
ADAHRS B	360	029	059	089	120	150	179	209	240	271	300	330
ESIS	001	030	060	090	120	150	180	210	240	270	300	331
Standby	358	025	052	086	119	150	179	206	240	270	300	330

Pressure Altitude Calibration has been completed in accordance with the applicable ground test procedures:

System	Serial Number	Calibration Date	Max Altitude Tested
ADAHRS	KSG7200-B3060	28-Nov-2018	25,000 ft
ESIS	2354	28-Nov-2018	25,000 ft
Transponder 1	4006	28-Nov-2018	25,000 ft
Transponder 2	4013	28-Nov-2018	25,000 ft

The **Basic Weight** and **Centre of Gravity** given below is based on weight and balance report AEROC V6.3.G.1 and has been performed in accordance with Viking Air Limited work instruction WI-017-14.

Basic Aircraft Weight (lbs):	Center of Gravity (inches):
7,204.00	213.01

A Pre-Flight Inspection Checklist has been completed in accordance with FORM-017-27 on the following date: — Date: 20-Mar-2019

Production Release

A list of applicable Airworthiness Directives can be found in Section 3 of the Airframe Technical Record.

The activities described above have been performed in accordance with the applicable Viking Air Limited Engineering Data and inspected as per Viking Air Limited Quality System.

Signature:	Date:	Stamp:
	MAR 2 9 2019	SOC 532 DELEGATE

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp: SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	7	5	5	5

Certification Activities

STATEMENT OF CONFORMITY (For a newly Manufactured Aircraft)

Aircraft

Manufacturer:	Model:	Serial Number:	Type Certificate:
Viking Air Limited	**DHC-6 Series 400**	**976**	**A-82**

Engines

	Manufacturer:	Model:	Serial Number:	Type Certificate:
1.	**Pratt & Whitney Canada Corp.**	**PT6A-34**	**PCE-RB1175**	**E-6**
2.	**Pratt & Whitney Canada Corp.**	**PT6A-34**	**PCE-RB1176**	**E-6**

Propellers

	Manufacturer:	Model:	Serial Number:	Type Certificate:
1.	**Hartzell Propeller Inc.**	**HC-B3TN-3D\T10282N**	**BUA34153**	**P-49**
2.	**Hartzell Propeller Inc.**	**HC-B3TN-3D\T10282N**	**BUA34244**	**P-49**

Certification Statement

This aircraft was satisfactorily test flown in accordance with the FTP-6-400 Production Flight Test Procedure:

Document Title:	Number:	Dated:
Production Flight Test Certificate	**FTP-6-CERT**	**11-Mar-2019**

I hereby certify that the aircraft described above, produced in accordance with the policies and procedures of the Viking Air Quality Manual (QM-001) under Transport Canada Certificate of Approval Number 22-80, conforms to its approved type design and is in a condition for safe operation.

Date: MAR 2 9 2019

Signature: _____
(On Behalf of Viking Air Limited)

SOC 532 DELEGATE

The above entry has been transcribed from the Aircraft Journey Logbook. Stamp:

SOC 532 DELEGATE

	MSN	TTSN	TCSN		
			#1 Engine	#2 Engine	Landings
Carried forward from the previous page ▶▶▶▶	976	7	5	5	5

Reserved for Transport Canada

Certificate of Airworthiness

This aircraft has been issued a Normal Category Certificate of Airworthiness as per Transport Canada Aviation Regulations CAR Part V subpart 507.02.

Minister Delegate – Production: J. GOCKSEDGE

Date: APR 0 2 2019

Stamp: DEPARTMENT OF TRANSPORT - MINISTÈRE DES TRANSPORTS — Délégué de l'Aviation civile — Canada — 0546 — Civil Aviation Delegate

Reserved for Transport Canada

Export Airworthiness Certificate

This aircraft has been issued a Canadian Export Airworthiness Certificate as per Transport Canada Aviation Regulations CAR Part V subpart 509.03.

Minister Delegate – Production:

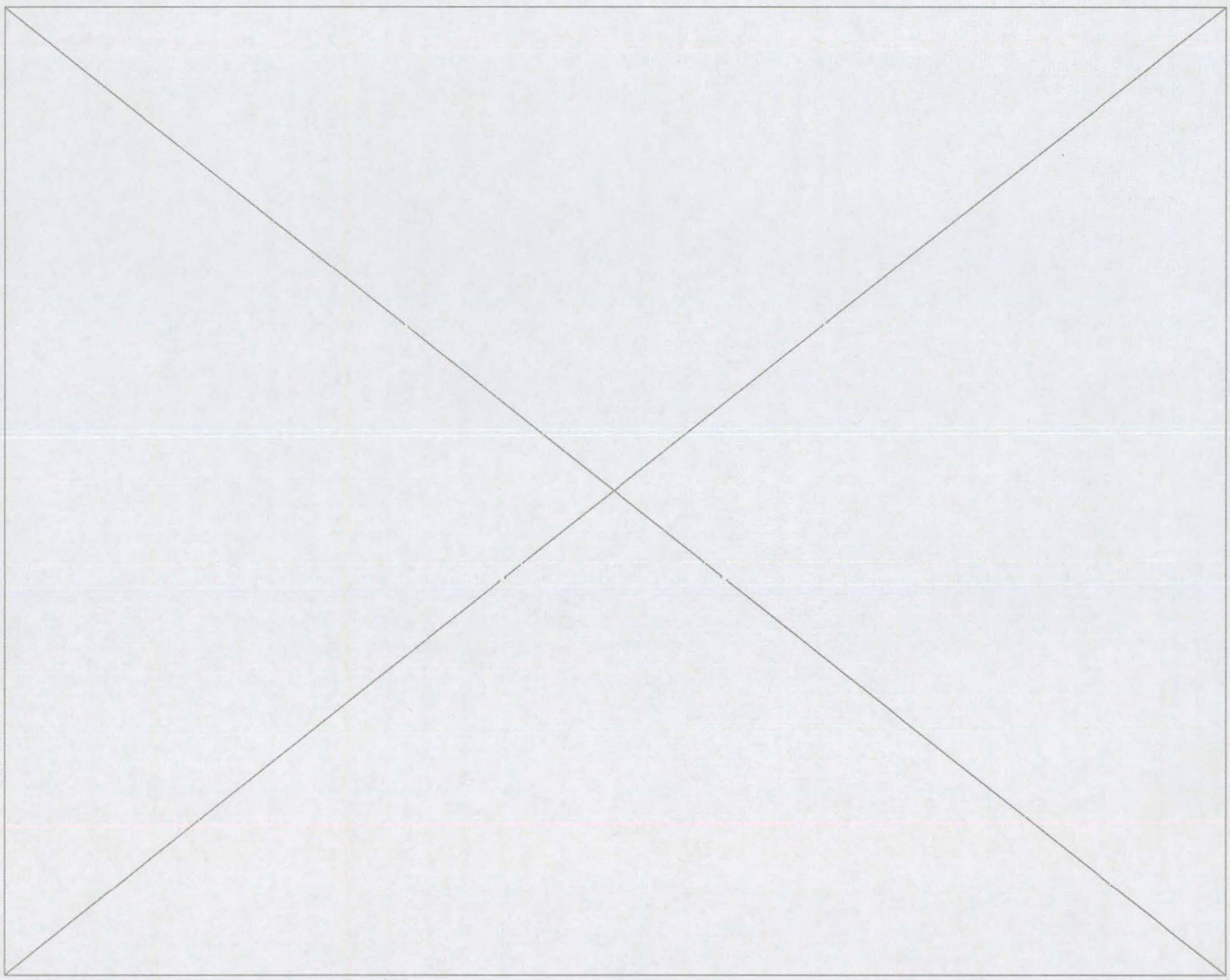
Date:

Stamp:

The above entry has been transcribed from the Aircraft Journey Logbook.

Stamp: SOC 532 DELEGATE

Model: DHC-6 Series 400				Serial Number: 976		
Date Installed	**Total Airtime**	**Component**	**Model**	**Serial Number**	**Position**	**Stamp**
13-Dec-2018	0.0	Engine	**PT6A-34**	**PCE-RB1175**	1	
13-Dec-2018	0.0	Engine	**PT6A-34**	**PCE-RB1176**	2	
13-Dec-2018	0.0	Propeller	**HC-B3TN-3D\T10282N**	**BUA34153**	1	
13-Dec-2018	0.0	Propeller	**HC-B3TN-3D\T10282N**	**BUA34244**	2	



N/A* - *Not applicable,* ***C* *– Complied with,* ***R*** *– Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
CF-68-05	Additional balance weights introduced to assure ailerons meet aerodynamic requirements.	MOD 6/1104 incorporated during manufacture.		X		
CF-68-09	Reinforcement to left hand rear Spar to preclude possibility of elevator trim tab flutter.	MODs 6/1120, 6/1135 and 6/1142 incorporated during manufacture.		X		
CF-69-20	To detect loose or missing Jo-bolts in wing struts which have accumulated 150 hours or more time in service.	Not applicable due to wing strut part number.	X			
CF-70-10	To provide adequate protection for particular circuit wires.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X			
CF-71-07	Improved sump plate flapper valve seal – To preclude separation of seal from flapper valve in cells 4 and 5.	MOD 6/1406 incorporated during manufacture.		X		
CF-71-11	To prevent possible explosions in bays 9 and 10 due to the operation of electrical components, deactivate the J.B. Systems Air Conditioner.	Not applicable air-conditioning system not installed.	X			
CF-71-12	Slow blow fuses Type 313 replaced with fast blow fuses Type 312 – To improve protection of AC power on aircraft.	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X			
CF-1972-06R5	Reinforced lower torque tube – To eliminate 50–hour inspection required by F.A.A. A.D. Redesigned Elevator/Rudder pulley bracket – to prevent buckling and cracking due to high ground gust loading.	MODs 6/1262, and 6/1433 incorporated during manufacture.			X	Recurring inspection of elevator quadrant due at 400 flight hours.
CF-73-03R1	Wing Rear Spar – Change of material provide improved stress corrosion cracking properties.	MOD 6/1301 incorporated during manufacture.		X		
CF-73-04	Propeller Autofeather – To introduce autofeather control on PT6A-20 engines	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X			

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
CF-73-06R2	Overhead console control quadrant – Introduction of improved material – To preclude cracking.	MOD 6/1895 incorporated during manufacture.		X		
CF-73-09	Propeller Autofeather – Pressure switches wiring modified to overcome erratic operation of single pole pressure switches.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X			
CF-73-11	Float attachment bolts – Special inspection and replacement	Not applicable floats not installed.	X			
CF-73-12	Side Seat Rail attachment – Special inspection of rivets and replacement as required.	Not applicable inward facing seats not installed.	X			
CF-74-04	Horizontal Stabilizer Rear Spar – Special inspection and repair.	Not applicable by aircraft serial number.	X			
CF-74-07	Electrical Power Cable below cockpit – Repositioning of terminal block 17.	Not applicable by aircraft serial number.	X			
CF-75-06	Slow blow fuses Type 313 replaced with fast blow fuses Type 312 – To improve protection of AC power on aircraft.	Not applicable due to aircraft serial number MOD 6/2028 incorporated during manufacture.	X			
CF-75-11	To preclude the possibility of partial or total electrical failure due to contact welding of reverse current relays, Hartman Part A700AAP or A700AP.	Not applicable new part number A701D installed. MOD 6/2028 incorporated during manufacture.	X			
CF-75-14	Floats – Applies to all de Havilland DHC-6 Model aircraft fitted with Canadian Aircraft Products Models 12000, 12000A or 12000B floats modified for water bombing operation	Not applicable floats not installed.	X			
CF-76-21	Lighting protection – Wing tip, fincap, nose fuselage (SFAR 23 requirement).	Not applicable by aircraft serial number. MOD 6/1513 installed during manufacture.	X			

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
CF-77-05R1	Elevator Bellcrank and Hand Pump support structure – To reinforce existing structure and provide additional support structure for hand pump.	MOD 6/1594 incorporated during manufacture.		X		
CF-77-08	To preclude the possibility of partial or total electrical failure due to contact welding of reverse current relays, Hartman Part A700AAP or A700AP.	Not applicable new part number A701D installed. MOD 6/2028 incorporated during manufacture.	X			
CF-78-17R1	Fuselage Side Frame, STA 218.125, and STA 219.525 – To alter material from 7079T651 to 7075T7351 for improved resistance to stress corrosion.	MODs 6/1461 and 6/1462 incorporated during manufacture.		X		
CF-79-02	Elevator Quadrant – Increased diameter bolts.	Not applicable MOD 6/1678 incorporated during manufacture.	X			
CF-79-22R2	Flight Control Push Rods – Subject rods are removed and replaced with improved rods of a different alloy. These modifications improve the inspection cycle to 2 years or 2400 hrs.	MOD 6/1781 incorporated during manufacture.			X	Recurring inspection due at 2400 flight hours or 2 years after the date of manufacture.
CF-79-24	Beta System – To introduce power lever test switch to enable pilot to check operation of beta throttle arming switch.	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X			
CF-80-01	Floats – Failures of the front spreader bar end lugs and inadequate clearances between fittings	Not applicable floats not installed.	X			
CF-80-03R4	Flight Control Push Rods – Subject rods are removed and replaced with improved rods of a different alloy. These modifications improve the inspection cycle to 2 years or 2400 hrs.	MODs 6/1785, 6/1791, 6/1792 and 6/1802 incorporated during manufacture.			X	Recurring inspection due at 2400 flight hours or 2 years after the date of manufacture.
CF-80-22	Applies to DHC-6 aircraft equipped with Pratt and Whitney PT6A-6B, PT6A-2/20 or PT6A-20 engines.	Not applicable PT6A-34 engines installed.	X			

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
CF-81-05	To prevent a possible electrical fire as a consequence of chafing causing a short circuit between wire looms and the co-pilot's dimming rheostat.	Not applicable due to aircraft serial number. MOD 6/2028 incorporated during manufacture.	X			
CF-82-20	Applies to all DHC-6 aircraft incorporating windscreen washer/de-icing system.	Not applicable specified deicing equipment not installed.	X			
CF-82-21R1	Elevator – To provide lateral stiffening of root rib and adjacent nose rib.	MOD 6/1769 incorporated during manufacture.		X		
CF-83-11R1	Elevator, Hinge Arm – To provide an improved lug to prevent cracking, galling and enlargement of bolt holes.	MOD 6/1799 incorporated during manufacture.		X		
CF-83-25R2	Applies to the De Havilland Aircraft of Canada DHC-6 Twin Otter Aircraft, all series, when equipped with cabin utility seats.	Not applicable utility seat configuration not installed	X			
CF-83-32	De Havilland Modification No. 6/1273 introduced a fuel vent flame arrestor for lightning protection purposes.	Not applicable MOD 6/1375 incorporated during manufacture.	X			
CF-84-03	Landing Gear – Nose wheel steering cables material change.	MOD 6/1800 incorporated during manufacture.		X		
CF-85-12R3	Wings – Front attachment adapters introduction of steel adaptor fittings.	MOD 6/1887 incorporated during manufacture.		X		
CF-85-14	Wings – Wing Struts special inspection for internal beam corrosion.	Not applicable by strut part number C6W1005-9/-10 installed during manufacture.	X			
CF-87-02R1	Wings – Special inspection of Main Spar lower cap angles.	Incorporated during manufacture.			X	Threshold inspection due 10 years after the date of manufacture.

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
CF-90-01	Gust Lock – Revision to lock control column in forward position and to introduce flag for positive indication of gust lock engagement.	MODs 6/1676 and 6/1726 incorporated during manufacture.		X		
CF-90-10	Elevator, Push Rods – To detect cracked rod ends or other anomalies, and prevent recurrence of rod end failure.	Not applicable new push rods installed during manufacture.	X			
CF-91-30	Wing Struts – Anti-chafing nylon shield added to strut to prevent damage due to chafing action of upper fairing.	MOD 6/1581 incorporated during manufacture.		X		
CF-92-04	Horizontal Stabilizer – Special inspection and rectification of forward and aft attachment brackets redesigned retention method.	Not applicable by aircraft serial number. MODs 6/1808, and 6/1809 incorporated during manufacture.	X			
CF-92-05	Horizontal Stabilizer – Special inspection and rectification of center hinge bracket for cracks.	Not applicable new parts installed during manufacture.			X	Recurring inspection due at 1200 flight hours.
CF-93-25	Wings – Lower skin and leading edge rivets at front spar inspection and replacement	MODs 6/1645 and 6/1893 incorporated during manufacture.		X		
CF-94-12R1 (See also CF-99-11)	Corrosion Inspection Tasks – accomplish all corrosion tasks specified in Part 3 of the Del lavilland DI IC-6 Twin Otter Corrosion Prevention and Control Manual (PSM 1-6-5 Rev.1 dated Dec. 16, 1998)	Incorporated during manufacture.		X		
CF-99-11	Corrosion Inspection Tasks – accomplish all corrosion tasks specified in Part 3 of the DeHavilland DHC-6 Twin Otter Corrosion Prevention and Control Manual (PSM 1-6-5 Rev.1 dated Dec. 16, 1998)	Incorporated during manufacture.			X	Inspections due as per PSM 1-6-5 Part 3
CF-2007-31	Wing – Forward Spar Adaptor C6WM1027-1 inspection and replacement.	Not applicable new part number Adaptor C6WM1027-3 incorporated during manufacture.	X			

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400		Serial Number: 976				
A.D. Number	**A.D. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
2012-26-15R1	Pressure measurement error in the air pressure transducer	Not applicable by component serial number.	X			
CF-2017-20	Flight Controls –Aileron Cable wear at wing root rib and at fuselage skin.	New aileron control cables installed during manufacture.			X	Initial inspection due prior to 300 flight hrs. Subsequent 4 inspections due at 500 flight hr. intervals
CF-2018-07	Fuel System – Corrosion of Fuel System Components	Not applicable due to aircraft serial number.	X			
CF-2018-28	Flight Controls – Worn Elevator Cables	New elevator control cable and pulley installed during manufacture.			X	For elevator control cables that have reached life limit inspect cable and pulley at STA 270.3.
CF-2019-02	Time Limits / Maintenance Checks – Airworthiness Limitations - Implementation	New parts installed during manufacture			X	Refer to PSM 1-6-11 Revision 9 for applicable inspection intervals and airworthiness limitations.
End of List						

As of this date, I certify that this aircraft complies with all applicable TCCA Airworthiness Directives Reference TCCA Airworthiness Directives through: **TCCA AD#: CF-2019-11**

IF applicable, additional foreign Airworthiness Directives have been appended to this section of the technical record.

None: ☑ Refer to the following pages: ☐

Signature:	Date: MAR 2 9 2019	Stamp: SOC 532 DELEGATE

N/A - Not applicable, **C – Complied with, **R** – Recurring Inspection required*

Model: DHC-6 Series 400						Serial Number: 976
S.B. Number	**S.B. Description**	**Method of Compliance**	**N/A***	**C***	**R***	**Interval**
---	All Mandatory Service Bulletins applicable to the DHC-6 Series 400 have been incorporated during manufacture.	---	---	---	---	---
End of List						

As of this date, I certify that this aircraft complies with the Viking Air Limited issued Mandatory Service Bulletins listed above.

Signature:	Date: MAR 2 9 2019	Stamp: SOC 532 DELEGATE

*N/A - Not applicable, **C** – Complied with, **R** – Recurring Inspection required*

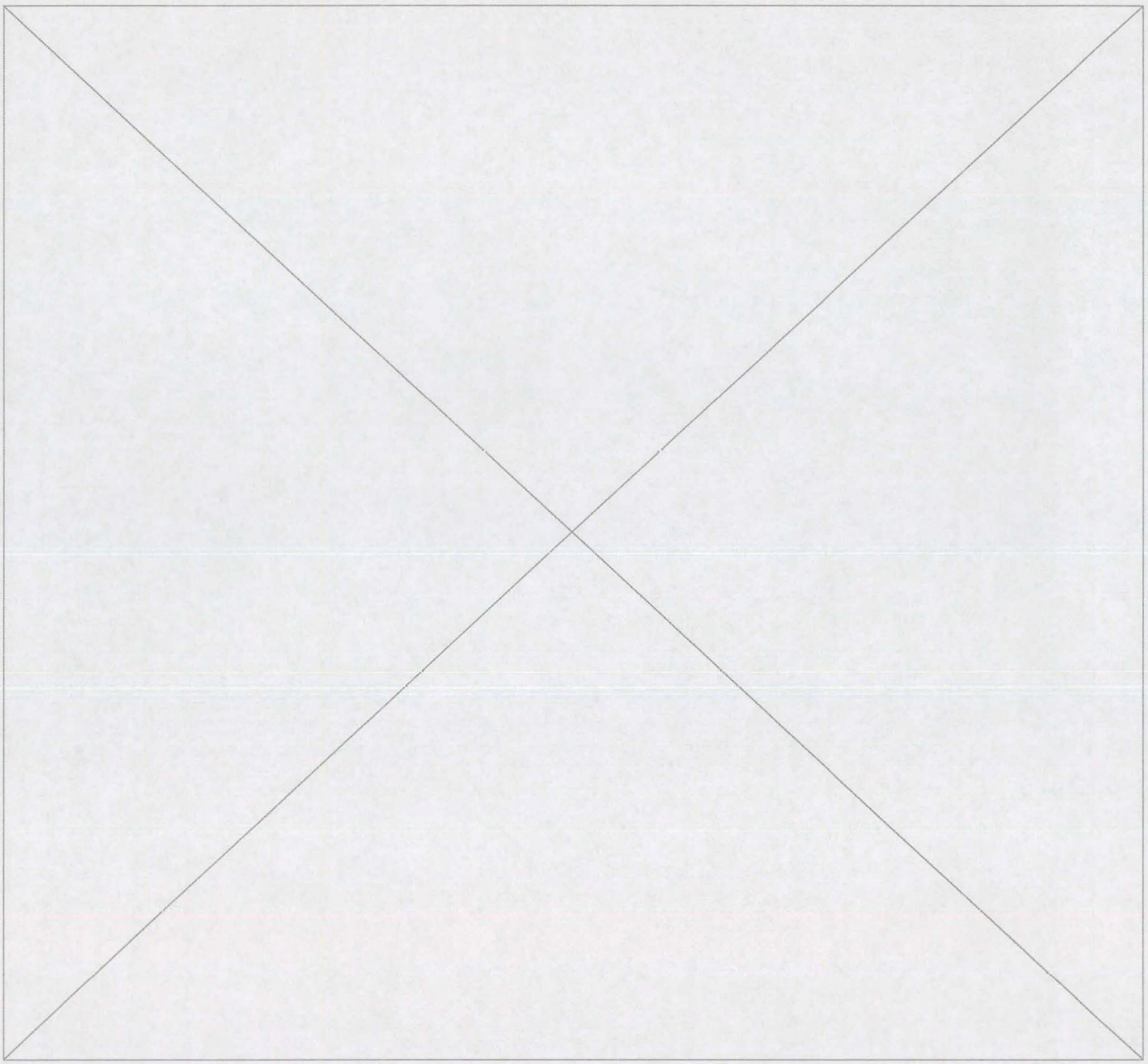
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Model: DHC-6 Series 400	Serial Number: 976

All modifications embodied during production of this aircraft are listed in the following document, which is appended to this section of the technical record:

Document Title:	Number:	Issue:
Production Configuration Statement	PCS-SN976	Issue 1

Signature:	Date:	Stamp:
	MAR 2 9 2019	SOC 532 DELEGATE



Production Configuration Statement
DHC-6 Series 400
Serial Number 976

The Modifications (MODs) and Standard Order Options (SOOs) listed below have been incorporated during the production of this aircraft (ref. PSM 1-6-12 MOD and OPTIONS Index). These MODs and SOOs, as well as any exceptions affecting aircraft maintainability, meet the requirements as defined by VAL Engineering in document:

• Engineering Configuration Statement ECS- 976 Issue 1

Production Configuration is defined by the following Sections:

1. Modifications (MODs)
2. Standard Order Options (SOOs) including any MODs required for this option
3. Non-Conformances affecting aircraft maintainability

SECTION 1: The following approved Modifications have been incorporated into production:

6/1003	ELEVATOR BALANCE WEIGHT
6/1004	WING HINGE ARMS
6/1005	BAGGAGE COMPARTMENT
6/1007	ELEVATOR INBOARD FAIRING
6/1008	RUDDER HINGE BRACKET ANGLE
6/1011	WING FLYING CONTROLS
6/1012	TAILPLANE CONFIGURATION
6/1013	RUDDER FAIRING
6/1021	COCKPIT IMPROVEMENT
6/1025	OVERHEAD CONSOLE THROTTLE LEVER
6/1028	ENGINE NACELLE
6/1029	PILOT AND COPILOT SEAT IMPROVEMENT
6/1030	ENGINE OIL TEMPERATURE INDICATION
6/1031	MAIN LANDING GEAR
6/1032	FUEL SYSTEM
6/1037	MAIN LANDING GEAR LOWER PLATEN ATTACHMENT BOLT
6/1038	AILERON BELLCRANK
6/1042	HYDRAULIC POWER PACKAGE
6/1043	ENGINE LOWER COWL
6/1044	FUEL SYSTEM, BOOST PUMP
6/1048	FUEL SYSTEM BOOST PUMP INTERCONNECT PIPING REPLACEMENT
6/1056	SNOW SEPARATOR IN ENGINE LOWER COWL
6/1059	STRUCTURAL FRAME STATION 239
6/1063	FUSELAGE FRAME
6/1065	PILOT STATIC SYSTEM
6/1066	WING FLAP HINGE ARM
6/1069	CABIN R.H. DOOR
6/1070	HEATING AND VENTILATION SYSTEM
6/1071	LANDING GEAR LEG FAIRING
6/1076	REAR BAGGAGE COMPARTMENT EXTENSION
6/1077	NOSE BAGGAGE COMPARTMENT EXTENSION
6/1079	COCKPIT NOISE REDUCTION
6/1084	FLAP HYDRAULIC PRESSURE

6/1085	FLAP CONTROL, CENTER FUSELAGE
6/1086	FUEL BOOST PUMP
6/1088	FUSELAGE FUEL SYSTEM
6/1090	ELECTRICAL SYSTEM- DELETE START CNTRL DELAY
6/1094	WING FRONT SPAR TOP CAP
6/1095	FLAP SELECTOR CONTROL VALVE
6/1098	MAIN LANDING GEAR FAIRING
6/1100	WING FLAP
6/1102	FUSELAGE, REAR
6/1103	FIN
6/1104	AILERON MASS BALANCE
6/1107	FLAP SELECTOR VALVE AND ACTUATOR
6/1108	ENGINE COWLING
6/1110	ELECTRICAL SYSTEM
6/1112	NOSE LANDING GEAR
6/1113	FUSELAGE CABIN SIDEWALL SEAT RAIL
6/1114	TAILPLANE TO FUSELAGE ATTACHMENT BOLT
6/1115	FLYING CONTROL SYSTEM
6/1117	WING (Main Spar)
6/1120	ELEVATOR TRIM TAB
6/1121	ELEVATOR/FLAP INTERCONNECT TAB
6/1122	AILERON TRIM TAB
6/1123	AILERON GEARED TAB
6/1124	RUDDER TRI TAB
6/1127	FLYING CONTROL IMPROVEMENT
6/1128	FLYING CONTROL IMPROVEMENT
6/1129	FLYING CONTROL IMPROVEMENT
6/1135	ELEVATOR
6/1142	ELEVATOR TOP SKINS
6/1143	ENGINE GEARBOX
6/1153	RUDDER FAIRING
6/1157	REAR BAGGAGE COMPARTMENT
6/1159	WING STRUT
6/1160	MAIN LANDING GEAR
6/1162	NACELLE ENGINE CONTROL BRACKET
6/1164	NOSE WHEEL STEERING CABLE
6/1171	ENGINE VIBRATION MOUNT
6/1172	WING STRUT
6/1173	WING STRUT ATTACHMENT
6/1174	WING STRUT TOP LINK
6/1176	CABIN SEATING
6/1179	LOWER ENGINE COWLING
6/1181	HEAT AND VENT SYSTEM
6/1182	ELEVATOR AND RUDDER HINGE CUT-OUT
6/1187	FLYING CONTROL RUDDER GEAR TAB
6/1189	FUSELAGE/WING FRONT FITTING ADAPTERS
6/1193	FUSELAGE
6/1194	FUSELAGE
6/1195	FUSELAGE AND WING SHROUD
6/1198	ENGINE CONTROL PULLEY ASSEMBLY
6/1201	FIRE DETECTION UNITS
6/1203	WING BOTTOM SKIN
6/1204	NOSE GEAR STEERING ACTUATOR
6/1206	FLYING CONTROL

6/1207	FLYING CONTROLS
6/1209	FLYING CONTROLS
6/1211	FUSELAGE STRUCTURE
6/1213	FLYING CONTROL
6/1214	FLYING CONTROL
6/1215	SNOW SEPARATOR
6/1219	FLYING CONTROLS, TRIM SYSTEM - RUDDER AND FLAP/ELEVATOR
6/1222	WING FENCES
6/1224	WING FLAPS
6/1225	CABIN INTERIOR
6/1226	POWER PLANT, OIL TEMPERATURE CONTROL FLAP ACTUATOR
6/1227	POWER PLANT, IMPROVED OIL TEMPERATURE CONTROL SYS
6/1228	HYDRAULIC RESERVOIR, DIPSTICK ASSEMBLY
6/1230	NOSE BAGGAGE COMPARTMENT
6/1231	NOSE WHEEL STEERING CONTROL PULLEY BRACKET
6/1232	CABIN R.H. DOOR
6/1233	MAIN LANDING GEAR
6/1235	RUDDER CONTROL, GUST LOCK MECHANISM REDESIGNED
6/1245	RUDDER STOP
6/1247	PISTON, NOSE LANDING GEAR
6/1248	WING FRONT SPAR ATTACHMENT
6/1250	NOSE LANDING GEAR STEERING ACTUATOR
6/1253	SAME AS MODIFICATION 6/1252 BUT FOR PT6A-27
6/1255	AVIONIC EQUIPMENT SHIELDING, NOSE BAGGAGE COMPARTMENT
6/1256	CABIN ROOF
6/1257	MAIN LANDING GEAR
6/1258	ENGINE NACELLE FIREWALL
6/1262	ELEVATOR/RUDDER PULLEY BRACKET
6/1263	ENGINE FIRE EXTINGUISHING SYSTEM
6/1264	NOSE BAGGAGE DOOR, STAY BAR
6/1268	DOOR OPEN WARNING
6/1269	CABIN DOOR R.H.
6/1270	COCKPIT DOOR LATCH MECHANISM
6/1272	LIGHTNING PROTECTION
6/1274	AC/DC POWER DISTRIBUTION SYSTEM
6/1277	TO INTRODUCE AUDIBLE STALL AND FIRE WARNING SYSTEM
6/1279	NOSE BAGGAGE COMPARTMENT, ELECTRICAL BONDING
6/1290	M.L.G. LEG FAIRING
6/1293	BLOCKING DIODE, EXTERNAL POWER SUPPLY
6/1294	WING FAIRLEAD BRACKET
6/1296	ENGINE COWLING, SNOW DEFLECTOR DOOR ASSY
6/1298	MAIN LANDING GEAR, AFT FAIRING
6/1299	NACELLE, POWER PLANT
6/1301	ATTACHMENT CAP, WING REAR SPAR
6/1302	NACELLE, COWL HOOK
6/1306	ENGINE EXHAUST
6/1308	MAIN LANDING GEAR
6/1309	FUEL SYSTEM
6/1315	HEAVY DUTY NICAD BATTERY
6/1316	ENGINE WASH SYSTEM, SPRAY RING
6/1319	NACELLE STRUCTURE REINFORCING
6/1322	RELOCATION OF T5 TEMPERATURE INDICATOR TRIM UNIT
6/1325	INTRODUCTION OF DOOR LOCK ON NOSE BAGGAGE COMPART
6/1326	ENGINE OIL BREATHER RELOCATION

6/1328	FLYING CONTROLS, GROUND LOCK
6/1329	PROPELLER AUTOFEATHER SYSTEM
6/1334	WING, OUTER HINGE ARMS
6/1335	FLAP, HINGE ARM ATTACHMENT
6/1336	FLAP, INBOARD FORE HINGE ARMS
6/1337	FLAP, INBOARD TRAILING HINGE ARMS
6/1339	FIN, HINGE ARMS
6/1340	RUDDER, HINGE BRACKETS
6/1341	TAILPLANE, HINGE ARMS
6/1342	ELEVATOR, HINGE ARMS
6/1344	M.L.G. LINK/PLATEN JOINT
6/1346	M.L.G. LINK/PLATEN JOINT
6/1348	AILERON, HINGE ARMS
6/1349	FLAP, OUTBOARD FORE HINGE ARMS
6/1351	RUDDER PEDAL ASSEMBLY, PUSH ROD REDESIGN
6/1352	HEAT AND VENT SYSTEM, FRESH AIR SCOOP REDESIGNED
6/1354	WING "D" NOSE DOOR ASSEMBLY
6/1360	BRAKE OPERATING LEVER, REDESIGNED
6/1361	LOWER COWL (TITANIUM PLATE REMOVAL)
6/1363	FLIGHT COMPARTMENT, PORT SIDE
6/1367	CAM - PROPELLER PITCH
6/1368	NOSE GEAR, HYDRAULIC LINES COVER
6/1375	LIGHTNING STRIKE PROTECTION
6/1376	RUDDER GEAR TAB
6/1390	FUSELAGE, REAR BAGGAGE COMPART
6/1395	AIRCRAFT OPERATING PLACARD
6/1396	NOSEWHEEL STEERING CONTROL
6/1397	RUDDER PEDAL
6/1403	LANDING LIGHT COVER
6/1406	FUEL SYSTEM
6/1407	AILERON CONTROL
6/1412	ENGINE CONTROLS, POWER/PITCH INTERLOCK LEVER
6/1415	REAR BAGGAGE COMPARTMENT FLOOR PANEL FASTENERS
6/1417	FUEL SYSTEM
6/1419	MAIN LANDING GEAR
6/1426	ENGINE VIBRATION MOUNTS
6/1427	WING, REAR SPAR LOWER SKIN JOINT
6/1433	CONTROL COLUMN
6/1435	FLYING CONTROL, AILERON CIRCUIT
6/1439	ENGINE CONTROLS
6/1448	HYDRAULIC RESERVOIR, VENT LINE
6/1451	OVERHEAD CONSOLE, FUEL CONTROL LEVERS
6/1453	ENGINE BREATHER HOSE
6/1458	ELEVATOR CABLE FAIRLEAD, REAR FUSELAGE FIN STRUCTURE
6/1461	FUSELAGE SIDE FRAME, STA 218, 125
6/1462	FUSELAGE SIDE FRAME, STA 219, 525
6/1463	FUSELAGE TOP BEAM, WING PICKUP
6/1468	POWER AND PROPELLER CONTROLS
6/1469	MAIN LANDING GEAR, COMPRESSION AND REBOUND BLOCKS
6/1470	AUTOFEATHER SYSTEM
6/1473	FIN - MOUNTED VOR/LOCALIZER ANTENNA
6/1477	WING REAR SPAR
6/1486	RUDDER CONTROL RODS
6/1494	REAR FUSELAGE

6/1496	WING TOP PANEL
6/1497	ENGINE COWLING REINFORCEMENT
6/1499	HYDRAULIC POWER PACK
6/1501	ELEVATOR TIP TRAILING EDGE
6/1502	PILOT CONTROL WHEEL
6/1505	PROPELLER RESTRAINT
6/1506	STATIC SELECTOR VALVE
6/1508	FUEL SYSTEM HOBBS SWITCHES
6/1513	STROBE NAVIGATION LIGHTS
6/1515	ENGINE CONTROLS
6/1523	FUEL PRESSURE SWITCH
6/1529	FLAP-INBOARD TRAILING-CENTER HINGE ARM
6/1530	FRONT FUSELAGE HYDRAULIC BAY ACCESS DOOR
6/1533	ENGINE SNOW SEPARATOR
6/1537	NOSE LANDING GEAR
6/1538	INSTRUMENT FRAME SHOCKMOUNTS
6/1540	WING SPAR, SKI TIE POINTS
6/1541	NOSE LANDING GEAR
6/1542	RELOCATION OF GRAVINER FIRE EXTINGUISHER MODEL 2
6/1543	HEAT AND VENT SYSTEM
6/1547	FUSELAGE JACKING POINT
6/1548	NOSE SPAR WEB
6/1554	NOSE LANDING GEAR
6/1556	INBOARD TRAILING FLAP HINGE REINFORCEMENT
6/1557	ENGINE MOUNT BUSH MORE BEARING ON TAPER
6/1561	AIRSTAIR DOOR QUICK RELEASE BRACKET
6/1562	N.L.G. CENTERING NOTCH
6/1563	RADIO SCREENS
6/1566	NOSE LANDING GEAR LOWER SUPPORT BEARING
6/1567	NOSE LANDING GEAR
6/1570	HYDRAULIC POWER PACKAGE
6/1580	NOSE LANDING GEAR
6/1581	WING STRUTS
6/1582	AIRSTAIR DOOR
6/1583	WASTE FUEL RETURN TO TANK
6/1585	REVERSE CURRENT RELAY REPLACEMENT
6/1590	VOLTAGE REGULATORS
6/1591	REVERSE CURRENT RELAYS RELOCATION
6/1593	CUFF ASSEMBLY - PITOT - STATIC SUPPORT
6/1594	ELEVATOR BELLCRANK AND HAND PUMP SUPPORT STRUCTURE
6/1595	INSTRUMENT PANEL VIBRATION ISOLATORS
6/1596	DEICING SYSTEM DETAIL CHANGES
6/1597	FIXED STRUCTURE UNDERCARRIAGE FAIRING DOUBLER RETENTIO
6/1599	ANTENNAS STRUCTURAL PROVISIONS
6/1600	TAXI LIGHT
6/1601	SHOULDER HARNESS AND INERTIAL REELS
6/1603	COPILOT'S STEP AND HANDGRIP
6/1604	COPILOT'S FLIGHT INSTRUMENT PANEL
6/1606	AIRSTAIR DOOR RESTRAINT
6/1609	FLAP SELECTOR POSITION INDICATOR LIGHT
6/1612	CIRCUIT BREAKER INSULATOR
6/1614	DE VORE BLEED AIR SYSTEM
6/1615	HORIZONTAL STABILIZER SKIN DOUBLER
6/1616	STEERING ACTUATOR LINE FILTER

6/1618	NOSE LANDING GEAR FORK NUT LOCKING
6/1620	MAIN LANDING GEAR
6/1633	CABIN AIR OUTLET VENT
6/1637	ENGINE LOWER COWL
6/1640	ENGINE COWLING BRACKET REDESIGN
6/1641	AIRSTAIR DOOR SWITCH IMPROVEMENT
6/1642	MAIN LANDING GEAR LEG
6/1643	FUSELAGE AFT OF FUEL TANKS - IMPROVED DRAINAGE
6/1644	WHEEL BRAKE PUSH ROD - INCREASED ADJUSTMENT
6/1645	WINGS
6/1650	FLAP ACTUATOR
6/1651	REVERSE CURRENT CIRCUIT BREAKER ELIMINATION
6/1655	NACELLE LONGERON
6/1660	MAIN LANDING GEAR LEGS
6/1668	CONTROL COLUMN AUTOPILOT CABLE CONDUITS
6/1669	WING MAIN SPAR
6/1672	MAIN LANDING GEAR
6/1673	ELEVATOR MASS BALANCE COVER PLATES
6/1675	ENGINE COWLING BRACKET
6/1676	GUST LOCK
6/1678	ELEVATOR QUADRANT ATTACHMENT
6/1680	BLEED AIR SYSTEM
6/1690	AILERON SERVO TAB LEVER
6/1691	DELETION OF REDUNDANT PIPE
6/1693	TAILPLANE SPACER BLOCK
6/1694	AILERON
6/1695	ELEVATOR
6/1696	RUDDER
6/1697	EMERGENCY LABEL
6/1701	RUDDER GUST LOCK
6/1702	WING TOP SKIN AND STRINGERS
6/1703	ELEVATOR CONTROL RODS
6/1704	AILERON CONTROL CABLE
6/1705	BAGGAGE COMPARTMENT LIGHT SWITCH
6/1709	CONTROL PULLEY INSTALLATION
6/1715	POWER LEVERS
6/1719	WING LOWER SKIN
6/1721	FLAP CONTROL ROD
6/1724	HORIZONTAL STABILIZER SKIN AND STRINGERS
6/1725	CONTROL COLUMN
6/1726	GUST LOCK
6/1732	NOSEWHEEL STEERING
6/1734	AILERON CONTROL ROD C6CW1018
6/1735	ELEVATOR TRIM AND ELEVATOR/FLAP INTERCONNECT CONTROL
6/1736	FLAP CONTROL RODS
6/1737	AILERON NOSE CAP
6/1738	ELEVATOR QUADRANT - REAR FUSELAGE
6/1739	ELEVATOR CONTROL ROD Addition of
6/1742	AILERON QUADRANT
6/1743	VERTICAL STABILIZER
6/1744	RUDDER PEDAL
6/1745	VERTICAL STABILIZER, LIGHTNING PROTECTION
6/1746	WING, LIGHTNING PROTECTION
6/1747	ELEVATOR CONTROL CABLES

6/1748	FLAP ACTUATOR FORWARD SUPPORT
6/1752	WING, REAR SPAR PICK
6/1753	AILERON, LEADING EDGES
6/1755	GUSTLOCK CONTROL, RUDDER
6/1759	AILERON CONTROL ROD/FLAP HINGE ARM
6/1760	FLIGHT COMPARTMENT DOORS
6/1761	PASSENGER CABIN DOORS
6/1762	LOWER WING SKIN
6/1763	NOSE LANDING GEAR
6/1764	MAIN LANDING GEAR
6/1766	REAR BAGGAGE COMPARTMENT
6/1769	ELEVATOR
6/1770	WING SHROUD
6/1773	AILERON SERVO TAB CONTROL
6/1774	ENGINE COWLING
6/1775	FLAP - ELEVATOR TRIM SYSTEM
6/1778	AIRSTAIR DOOR
6/1781	FLAP CONTROL RODS IN WING AND FUSELAGE
6/1785	ELEVATOR TRIM AND FLAP INTERCONNECT CONTROL RODS
6/1787	RUDDER GUSTLOCK CONTROL RODS
6/1788	WING INSPECTION DOOR STAY
6/1789	BRAKE VALVE PUSH RODS
6/1790	MAIN LANDING GEAR STAY TUBE ASSEMBLY
6/1791	AILERON CONTROL ROD
6/1792	ELEVATOR CONTROL ROD
6/1795	FUEL SYSTEM, BOOST PUMP
6/1797	ELEVATOR AND RUDDER TRIM JACKS
6/1798	ELEVATOR, RUBBER STOPS
6/1799	ELEVATOR, HINGE ARM
6/1800	NOSEWHEEL STEERING CABLES
6/1802	RUDDER TRIM
6/1806	MAIN LANDING GEAR
6/1807	FLYING CONTROLS - AILERON
6/1808	HORIZONTAL STABILIZER
6/1809	HORIZONTAL STABILIZER
6/1811	ENGINE MOUNTING BOLT
6/1812	TRAILING FLAP CONNECTING ROD
6/1823	ENGINE CONTROLS - FUEL CONTROL UNIT CAM
6/1838	ENGINE CONTROLS - CONSOLE POWER LEVER
6/1841	FUSELAGE/STABILIZER
6/1842	FUSELAGE/STABILIZER
6/1843	FUSELAGE/STABILIZER
6/1844	FUSELAGE/STABILIZER
6/1845	FUSELAGE/STABILIZER
6/1846	FUSELAGE/STABILIZER
6/1848	ICE AND RAIN PROTECTION
6/1850	FLIGHT CONTROLS
6/1858	IMPROVED N.L.G. LOWER BEARING
6/1859	ADAPTER FITTING - WING FRONT SPAR TO FUSELAGE
6/1866	FLAP TRIM SCREW JACK
6/1867	ENGINE
6/1868	WING
6/1870	WING
6/1871	WING

6/1873	WING
6/1877	FLAPS CABLE
6/1880	WING
6/1881	FUSELAGE
6/1884	WING
6/1887	WING FRONT SPAR TO FUSELAGE
6/1893	WINGS
6/1895	OVERHEAD CONSOLE ENGINE CONTROLS
6/1899	INTRODUCTION OF ALTERNATE FUEL HOSE ASSY
6/1902	MATERIAL SUBSTITUTION FOR STEP ASSEMBLY, COCKPIT
6/1908	DUCT INSTALLATION - PART NUMBER REVISION
6/1916	CONTROL CABLES
6/2000	SNOW SEPARATOR
6/2001	STA 60 BULKHEAD
6/2002	WHEELS & BRAKES
6/2003	ANTENNAS
6/2004	AIRCRAFT STRUCTURE
6/2005	FWD AVIONICS EQUIPMENT SHELVES
6/2006	FUEL SYSTEM
6/2007	AIRCRAFT STRUCTURE
6/2008	FLIGHT COMPARTMENT
6/2009	FUEL SYSTEM
6/2010	FRONT FUSE
6/2011	FORWARD CABIN
6/2012	PT6A-34 ENGINE (SERIES 400)
6/2013	AIRCRAFT INTERIOR
6/2014	CLIMATE CONTROL
6/2015	HORIZONTAL STABILIZER
6/2016	AIRFRAME MATERIALS & PROCESSES
6/2017	ELECTRICAL SYSTEM
6/2019	EXHAUST STACKS
6/2020	LONG NOSE MATERIAL CHANGE
6/2021	FAIRINGS
6/2022	PROPELLER (SERIES 400)
6/2023	FLAP CONTROL ROD SUPPORT ROLLER
6/2024	COMPOSITE DOORS
6/2025	FUSELAGE CROWN
6/2026	FUSELAGE
6/2027	CABIN FLOOR PANELS
6/2028	ELECTRICAL SYSTEM (SERIES 400)
6/2030	NOSE LANDING GEAR
6/2032	WINDSHIELD
6/2033	PITOT STATIC SYSTEM
6/2034	COCKPIT
6/2035	FUSELAGE STRUCTURE
6/2036	AIRFRAME
6/2037	BLEED AIR SYSTEM
6/2038	AIRCRAFT STRUCTURE
6/2039	HYDRAULIC SYSTEM - POWER PACK
6/2040	AIRFRAME
6/2041	COCKPIT STRUCTURE
6/2043	AIRFRAME
6/2044	AIRCRAFT STRUCTURE
6/2046	FUEL SYSTEM

6/2047	FLIGHT CONTROLS
6/2048	AFT AVIONICS EQUIPMENT SHELVES
6/2049	AVIONICS - APEX
6/2050	CABIN SEATING
6/2051	AIRFRAME
6/2052	COCKPIT
6/2053	BLEED AIR SYSTEM
6/2054	CABIN SEATING
6/2055	FLAP SELECTOR CONTROL VALVE
6/2056	NOSE LANDING GEAR
6/2059	MARKINGS AND PLACARDS
6/2060	FIRE EXTINGUISHER
6/2062	AIRFRAME
6/2063	AIRCRAFT STRUCTURE
6/2064	AIRCRAFT STRUCTURE
6/2066	WING TIP FUEL TANKS
6/2067	FORWARD CABIN
6/2068	FLIGHT CONTROLS
6/2070	STA 60 BULKHEAD
6/2071	AIRFRAME
6/2072	FORWARD CABIN
6/2073	WING TIP FUEL TANKS
6/2074	AIRCRAFT STRUCTURE
6/2075	FLIGHT COMPARTMENT
6/2076	FLIGHT COMPARTMENT
6/2077	FORWARD CABIN
6/2078	FORWARD CABIN
6/2079	FORWARD CABIN
6/2080	FORWARD CABIN
6/2081	CABIN
6/2082	WING TIP FUEL TANKS
6/2084	FLIGHT COMPARTMENT
6/2085	FLIGHT COMPARTMENT - POH STOWAGE
6/2086	FLIGHT CONTROLS
6/2087	FUSELAGE - COCKPIT STEP
6/2088	WINDSCREEN WIPER INSTALLATION
6/2089	AIRFRAME MATERIALS & PROCESSES
6/2090	VALIDATION OF VAL EOS
6/2091	FORWARD CABIN
6/2092	FIRE EXTINGUISHER
6/2093	HYDRAULIC SYSTEM - HANDPUMP
6/2094	AIRCRAFT STRUCTURE
6/2100	GROUND EQUIPMENT INCL. LAPTOP KIT SERIES 400
6/2101	INTRODUCTION OF ALTERNATE HARDWARE
6/2103	HORIZONTAL STABILIZER FAIRING ASSY
6/2104	HARDWARE UPDATES FOR LED LIGHTING SYSTEM
6/2105	UPPER TRANSPONDER ANTENNA
6/2106	STA 60 BULKHEAD STRUCTURE
6/2107	FWD & REAR AFT BAGGAGE COMPARTMENT LIGHT
6/2109	AILERON ROD ASSY
6/2110	AMETEK FUEL CONDITIONER - ELIMINATE CAS MESSAGE
6/2111	CLAMPING PROVISIONS STN 62.4
6/2112	WINDSHIELD
6/2113	CENTRAL CONTROL COLUMN

6/2114	AUDIO SHIELD WIRING
6/2115	AIRCRAFT INTERIOR
6/2116	STA 262.0 CABIN ROOF FRAME ASSY
6/2117	COVER
6/2118	ENGINE COWLS
6/2119	GEN FIELD RELAY PANEL ANGLE
6/2120	LOWER COWL DRAIN HOLES
6/2121	POSITION AND STROBE LIGHT SPLICES
6/2122	RADAR ALTIMETER TEST PINS
6/2125	OIL CAPACITY LABEL
6/2126	INTRODUCTION OF CONNECTOR BONDING TO AIRFRAME
6/2127	EQUIPMENT & FURNISHINGS
6/2128	COCKPIT DOOR
6/2129	FUSELAGE
6/2130	CIRCUIT BREAKER REPLACEMENT
6/2131	WEATHER RADAR ANTENNA INSTL
6/2133	NACELLE FIREWALL ASSY
6/2137	EQUIPMENT/FURNISHINGS - CARGO COMPARTMENT
6/2138	EQUIPMENT/FURNISHINGS - GENERAL
6/2139	REMOVAL OF BATTERY DRAIN HOLE
6/2140	INTRODUCTION OF LED WING INSPECTION LIGHT
6/2141	EQUIPMENT AND FURNISHINGS
6/2142	EQUIPMENT AND FURNISHINGS
6/2143	EQUIP. & FURNISHINGS, CARGO COMP., BLKHD 332, BAGGAGE
6/2145	INERTIAL SEPARATOR MOUNTING BRACKET AND DEFLECTOR
6/2146	EQUIPMENT AND FURNISHINGS
6/2147	FWD VHF MOUNTING STRUCTURE
6/2148	HANDWHEEL GRIP REPLACEMENT
6/2149	COVER ASSEMBLY, LANDING LIGHT
6/2150	TCAS I ANTENNA DOUBLER CUT
6/2151	FUSELAGE - PRODUCTION IMPROVEMENTS STN 200.36 W/L 44.0
6/2152	FORWARD AND REAR FIRE SEAL
6/2154	LANDING LIGHT COVER GASKET REINFORCING
6/2155	EQUIPMENT/FURNISHINGS - LOWER STN 332 BAGGAGE RESTRAINT
6/2163	BLEED AIR VALVE REPLACEMENT
6/2164	CONTROL COLUMN HOOD ASSY
6/2167	ELECTRONIC DISPLAYS - CHASSIS GROUNDING
6/2168	INTRODUCTION OF DRAIN TUBE TO AIR INTAKE DUCT FAN
6/2169	COCKPIT AREA MICROPHONE
6/2170	ESIS SPECIFICATION CHANGE (L3 GH-3900.2)
6/2171	ADAHRS MAGNETOMETER RE-LOCATION
6/2172	TORQUE TUBE
6/2180	HIRF COVER & RADIO EQUIPMENT COVER PRODUCTION IMPROVE
6/2186	FLAP ROD ROLLER INSTALLATION AT WING ROOT
6/2188	REMOVAL OF APEX HIGH SPEED ETHERNET BUS
6/2189	LANDING GEAR
6/2195	WASHER INSTL FORE & TRAILING FLAP INSTALLATION
6/2198	INTERIOR PLACARD IMPROVEMENTS
6/2201	BELLCRANK, C/S FLAP SYSTEM
6/2208	PITOT HEAD - INCREASED CORROSION PROTECTION
6/2209	INTRODUCTION OF ADDITION FUEL DRAINS AS STANDARD
6/2210	HYDRAULIC PRESS. TRANSDUCER BACKSHELL FOULING CORR.
6/2211	FLOOR PANEL SEAT TRACK COVER PLATE
6/2213	MOUNTING HARDWARE UPGRADE - MFD CONTROLLER HOUSING

6/2218	PRODUCTION IMPROVEMENTS TO OIL BREATHER INSTALLATION
6/2220	C6CF1297 RUDDER PEDAL SUPPORT ASSY
6/2221	ROOF FRAMES, STATION 143.95
6/2223	WARM AIR DISTRIBUTION IMPROVEMENTS
6/2224	HONEYCOMB CORE REPLACEMENT
6/2229	INTRODUCTION OF ANGLE TO ALLEVIATE FOULING CONDITIONS
6/2230	PRODUCTION IMPROVEMENTS TO NACELLE FUEL LINE INSTL
6/2234	ALTERNATE SWITCH ACTUATOR ASSY
6/2235	AILERON CABLE FAIRLEAD RELOCATION - WING ROOT
6/2236	ALTERNATE CHECK VALVE (CRISSAIR)
6/2238	INTRODUCTION OF WELDED GATE ASSY IN LIEU BRAZING
6/2239	BACK EMF SUPPRESSION TO MAU GIO CARD
6/2240	REPLACEMENT OF RED BEACON LIGHT
6/2241	REPLACEMENT OF A6693A UTILITY MOUNTING BASE
6/2251	PRODUCTION IMPROVEMENTS TO COCKPIT DOOR SURROUND
6/2254	STA 60.0 BULKHEAD LOWER STIFFENER, RIVET IN JOGGLE
6/2255	INSTRUMENT PANEL CENTRE COLUMN ACCESS IMPROVEMENT
6/2256	CVR & FDR INERTIA SWITCHES
6/2257	INTRODUCTION OF C6EC1002-97
6/2258	RAD ALT ANTENNA UPGRADE
6/2259	REMOTE ANGLE TO SEAT RAIL INSTALLATION
6/2260	TEXT UPDATE TO AVIONICS CIRCUIT BREAKER PANEL OVER
6/2261	FUEL SUMP PLATE MOUNTING BOLTS, LENGTH INCREASE
6/2262	FITMENT IMPROVEMENT TO RH OIL BREATHER LINES
6/2264	INTRODUCTION OF VSC7B4-0170 HOSE
6/2265	INTRODUCTION OF C6FS2638-13 ACCESS COVER
6/2272	INTRODUCTION OF NEW 15S STRINGER
6/2274	INTRODUCTION OF MACHINED SEAT PEDESTAL
6/2277	IMPROVEMENTS TO C6FS1495 BRACKET, FLAP HINGE
6/2278	AIRSTAIR DOOR AFT CABLE BRACKET
6/2279	OIL TEMP PROBE GASKET REPLACEMENT
6/2280	ALTERNATE RH SINGLE SEAT MOUNTING PROVISIONS
6/2283	IMPROVEMENT TO STA 111 BULKHEAD INTERIOR
6/2284	REROUTING OF IGNITION LEADS
6/2285	INTRODUCTION OF LED LANDING LIGHTS
6/2286	IMPROVEMENTS TO THUNDER BAY DOOR HANDLE
6/2294	INTRODUCTION OF TEAM JAS NOSE WHEEL
6/2295	TEAM JAS AIRSTAIR DOOR
6/2297	OIL COOLER DUCT MOUNTING BRACKET
6/2299	INTRODUCTION OF ALTERNATE FUEL PUMPS
6/2300	UNIVERSAL FIRE EXTINGUISHER PLACARD
6/2306	AUTO FEATHER ARMED SWITCH
6/2308	ALTERNATIVE EXIT SIGN
6/2310	AILERON CABLE COVER PROFILE MODIFICATION
6/2311	IMPROVED TAXI LIGHT ADAPTER
6/2313	HYDRAULIC POWER PACK - REPLACEMENT PRESSURE SWITCH
6/2314	INDEPENDENT GROUND & BATTERY SWITCHES
6/2315	ENGINE STARTING AVIONICS BUS
6/2316	EXTERNAL POWER AND BATTERY RELAYS
6/2319	ALTERNATE PROPELLER OVERSPEED GOVERNOR
6/2320	HEAT & VENT SYSTEM
6/2321	ELECTRICAL SYSTEM
6/2322	VERTICAL STABILIZER – RIVET INTERFERENCE
6/2325	ROUTING IMPROVEMENT OF CONTROL CABLE CONDUITS STA 111 B

6/2326	AIRSTAIR DOOR DESIGN IMPROVEMENT
6/2331	INTRO OF NLG-STEERING COLLAR - FRICTION DISK & SHIMS
6/2334	IMPROVEMENT OF AIRSTAIR DOOR SPRING CONTACTS
6/2339	MARKER BEACON SYSTEM
6/2340	INTRODUCTION OF RUBBER WASHERS
6/2341	COCKPIT DOORS
6/2343	PILOT AND CO-PILOT SEATS
6/2344	ENGINES - MODIFICATION TO FWD AND AFT FIRESEAL
6/2345	MANUFACTURER'S LABEL CLARIFICATION
6/2347	ENGINE CONTROLS -LEVER ASSY REVISION
6/2351	WING TIPS – ADDITION OF RUBBER WASHERS UNDER ATTACHING
6/2352	PRODUCE MACHINED RUDDER PEDALS
6/2357	STA 111.0 INTERIOR PANEL – DESIGN IMPROVEMENT (C6FF2973)
6/2359	ELECTRICAL POWER FEEDER CABLE
6/2364	INCORPORATION OF IMPROVED ENGINE COOLING AIR DUCT
6/2380	RED FUEL EMERGENCY SWITCHES
6/2382	UPDATE TP AGM-200 ADVANCED GRAPHIC MODULES
6/2383	ELECTRICAL WIRING
6/2384	ALTERNATE FCU LEVER
6/2385	INTRODUCTION OF ADAHRS PN 065-00188-5104
6/2386	SPRING, AIRSTAIR DOOR GUARD ASSY
6/2387	INTRODUCTION OF RUBBER WASHERS
6/2389	FUEL TANK MANIFOLD - SEALANT & ELECTRICAL BONDING
6/2390	FUEL PROBE ELECTRICAL CONNECTOR AND SEALING
6/2393	DRAIN VALVE
6/2394	IMPROVED ELASTIC BUNGEE CORD, MLG
6/2395	FUEL QUANTITY INDICATION PROBES
6/2403	BRACKET ASSY,CORROSION RESISTANCE IMPROVEMENT
6/2404	DOUBLERS,CORROSION RESISTANCE IMPROVEMENT
6/2406	ALTERNATE 14VDC POWER CONVERTER
6/2408	FUEL GALLERY SEALING AUGMENTATION
6/2414	CABIN AIR (RAM) CIRCULATION IMPROVEMENT
6/2417	ELEVATOR POTENTIOMETER ACCESS PANEL
6/2419	IMPROVED FLAP SELECTOR STOP
6/2423	EYE HEIGHT REFERENCE DEVICE – ORIENTATION CORRECTION
6/2433	IMPROVED HINGED FUEL CAP
6/2434	OPERATING INSTRUCTIONS PLACARD
6/2441	CABIN INTERIOR AESTHETIC IMPROVEMENTS
6/2450	INTRODUCTION OF KDU-1080-2 DISPLAY UNITS
6/2461	REPLACEMENT OF REAR BAGGAGE DOOR LATCH HANDLE
6/2466	REDUNDANT FEATURES REMOVAL FROM C6FS2619-29 INNER SKIN
6/2476	PT6A-27/-34 PROP GOVERNOR

SECTION 2: The following approved Options and the MODs required for the Option have been incorporated into production:

SOO 6205 CABIN EMERGENCY LIGHTING SYSTEM

SOO 6209 PRIVATE IFR OPERATIONS BUNDLE (NOW STD CONFIG)

SOO 6210 PUBLIC COMMERCIAL OPERATIONS BUNDLE (NOW STD CONFIG)

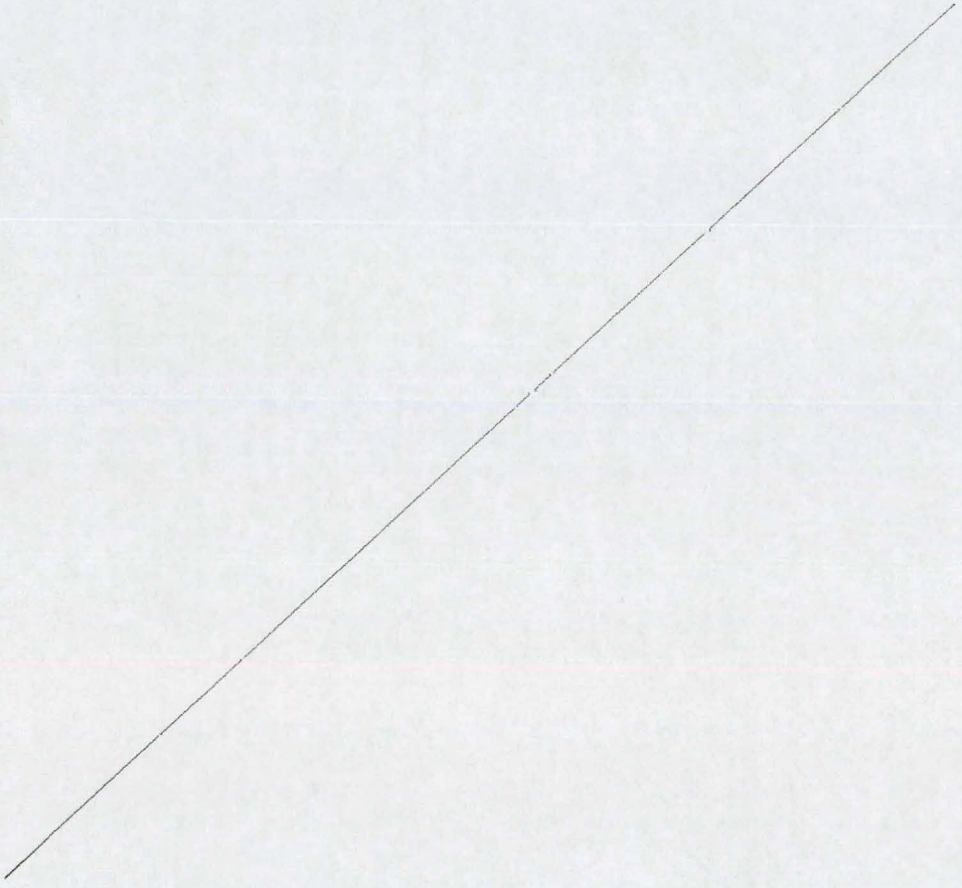
SOO 6230 ROSEN SUNVISOR INSTALLATION

SOO 6267 FLOAT REINFORCEMENTS – STA 111.0

SOO 6297 CONFIGURATION FOR CUSTOM SEAT ARRANGEMENT

SOO 6373 ALTERNATE NON-LOCKING HINGED FUEL CAP

6/2433 IMPROVED HINGED FUEL CAP



SECTION 3: The following non-conformances have been approved by the Material Review Board and affect the maintenance, limitations or performance of the aircraft:

NONE

T Eilkirk
Configuration Management

31 JAN 2019
Date